

GUJARAT TECHNOLOGICAL UNIVERSITY**BE - SEMESTER-V (NEW) - EXAMINATION – SUMMER 2016****Subject Code:2150101****Date:21/05/2016****Subject Name:Flight Mechanics****Time:02:30 PM to 05:00 PM****Total Marks: 70****Instructions:**

1. Attempt all questions.
2. Make suitable assumptions wherever necessary.
3. Figures to the right indicate full marks.

- Q.1** (a) With neat sketch explain V-n diagram. **07**
(b) Derive formula to Calculate Thrust required for level, unaccelerated flight. Also derive aerodynamic condition for minimum thrust required. **07**
- Q.2** (a) Define the pressure coefficient and explain the Prandtl-Glauert rule in detail along with a plot. **07**
(b) Explain the critical pressure coefficient and critical Mach number for different airfoils along with a plot. **07**
- OR**
- (b) Write a short note on: **07**
1. Explain flow separation
2. Significance of standard atmosphere in flight mechanics
- Q.3** (a) Discuss types of drag and explain Induced Drag in detail. **07**
(b) Describe in detail the advantages of trailing edge flaps. **07**
- OR**
- Q.3** (a) Derive Breguet Range formula. Also list out important points to maximize the range for reciprocating engine, propeller driven airplane. **07**
(b) Explain Static stability and dynamic stability. **07**
- Q.4** (a) Explain with neat sketches a short note on Mach cone. **07**
(b) Estimate the liftoff distance at sea level for the aircraft having total weight of 19815 N. Assume that $C_{L_{max}}$ is 1. Also when the airplane is on the ground the wings are 1.83 m above the ground. Wing Span is 16.25 m. Wing area is 29.54 m². $C_{D,0} = 0.022$ & $\mu_r = 0.02$. **07**
- OR**
- Q.4** (a) Explain Directional Static Stability. **07**
(b) Which are the necessary criteria for longitudinal balance and static stability? Explain in detail. **07**
- Q.5** (a) Derive formula to calculate Lift off distance. **07**
(b) Derive equation to calculate steady Rate of Climb. **07**
- OR**
- Q.5** (a) Explain Lateral Static Stability. **07**
(b) Write a short note on Neutral Point. **07**
