Seat No.:	Enrolment No.

GUJARAT TECHNOLOGICAL UNIVERSITY

BE – SEMESTER – V (NEW) EXAMINATION – WINTER 2015

Subject Code: 2150101 Subject Name: Flight Mechanics Time: 10:30am to 1:00pm Instructions:			Date:17/12/ 2015 Total Marks: 70	
		0:30am to 1:00pm Total Marks:		
	1. 2. 3.			
Q.1	(a)	Obtain lift coefficient from pressure coefficient. Plot and explain the pressure distribution over the upper and lower surface of an airfoil for small angles of attack.	07	
	(b)	Derive equations of Motion for an Airplane in translational motion. Reduce the same for level, unaccelerated flight.	07	
Q.2	(a)	Derive formula to Calculate Power required for level, unaccelerated flight. Also derive aerodynamic condition for minimum power required.	07	
	(b)	Explain Static stability and Dynamic stability. OR	07	
	(b)	Explain Critical Mach number and Drag divergence Mach number	07	
Q.3	(a)	 Explain compressibility corrections Distinguish between finite and infinite wings. 	07	
	(b)	Write a short note on Swept wings	07	
Q.3	(a) (b)	Write a short note on Neutral Point. Which are the necessary criteria for longitudinal balance and static stability? Explain in detail.	07 07	
Q.4	(a)	Estimate the Landing ground roll distance at sea level for the aircraft having empty weight of 54966 kg. No thrust reversal is used; However, spoilers are operated such that L=0. The maximum lift coefficient ($C_{L\ max}$) with fully flaps employed at touch down is 2.5. Wing area is 29.54 m ² . $C_{D,0} = 0.022$ & μ_r =0.4.	07	
	(b)	Derive formula to calculate Lift off distance. OR	07	
Q.4	(a) (b)	Derive and explain Wave drag. Explain Lift Augmentation	07 07	
Q.5	(a) (b)	Write a short note on Accelerated Rate of climb using Energy Method. With neat sketch explain V-n diagram.	07 07	
Q.5	(a) (b)	Explain Directional Static Stability. Explain Lateral Static Stability.	07 07	
